PHYSICS-INFORMED SELF-GUIDED DIFFUSION MODEL FOR HIGH-FIDELITY SIMULATIONS

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Abstract

Machine learning (ML) models are increasingly explored in fluid dynamics as a promising way to generate high-fidelity computational fluid dynamics data more efficiently. A common strategy is to use low-fidelity data as computationalefficient inputs, and employ ML techniques to reconstruct high-fidelity flow fields. However, existing work typically assumes that low-fidelity data is artificially downsampled from high-fidelity sources, which limits model performance. In real-world applications, low-fidelity data is generated directly by numerical solvers with a lower initial state resolution, resulting in large deviations from high-fidelity data. To address this gap, we propose *PG-Diff*, a novel diffusion model for reconstructing high-fidelity flow fields, where both low- and highfidelity data are generated from numerical solvers. Our experiments reveal that state-of-the-art models struggle to recover fine-grained high-fidelity details when using solver-generated low-fidelity inputs, due to distribution shift. To overcome this challenge, we introduce an *Importance Weight* strategy during training as self-guidance and a training-free Residual Correction method during inference as physical inductive bias, guiding the diffusion model toward higher-quality reconstructions. Experiments on four 2D turbulent flow datasets demonstrate the effectiveness of our proposed method. Code and further details are available here.

1 INTRODUCTION

High-fidelity simulations of computational fluid dynamics (CFD) are crucial for understanding fluid interactions in engineering systems, greatly impacting design and application outcomes (Wang et al., 2024; McGreivy & Hakim, 2024). Traditional approaches such as Direct Numerical Simulation (DNS) (Orszag, 1970) offer high-resolution solutions. However, they are computationally expensive, especially for complex dynamics such as turbulence with high Reynolds numbers (Zhang et al., 2023). Therefore, learning neural-based simulators from data become attractive alternatives, balancing between efficiency and simulation fidelity (Huang et al., 2023).

One popular strategy is to reconstruct high-fidelity data from low-fidelity inputs, which usually reduces the discretization grid size in the spatial domain to improve computational efficiency (Shu 040 et al., 2023; Pradhan & Duraisamy, 2021). Various machine learning models, including those based 041 on Convolutional Neural Networks (CNNs) (Fukami et al., 2019), Generative Adversarial Networks 042 (GANs) (Li & McComb, 2022), and Diffusion Models (Shu et al., 2023), have been developed to 043 reconstruct high-fidelity CFD data from low-fidelity inputs. Majority of them are categorized as 044 direct mapping models, which require both low- and high-fidelity data for training and can only capture the relationship between particular low-fidelity and high-fidelity pairs. In contrast, diffusion models only require high-fidelity data during training and can reconstruct from out-of-distribution 046 low-fidelity data during inference, by treating them as intermediate samples in the denoising stage. 047

Motivation: one fundamental drawback in existing studies is that they assume low-fidelity data is ar tificially downsampled from high-fidelity data at the same timestamp. Such data inherently has more
 information compared to solver-generated low-fidelity data in reality, where coarser discretization
 grids are used in numerical solvers for saving computational resources. As illustrated in Figure 1,
 the former follows "integrate then downsample", which starts from the high-fidelity initial states to
 rollout trajectories, and then downsample at each timestamp. The latter follows "downsample then
 integrate", which downsamples the high-fidelity initial state to obtain coarser discretization grids

as starting points and rollout trajectories through numerical solvers. Therefore, models trained in
the downsampled settings can lead to inferior performance during inference when given the solvergenerated data in reality, as solver-generated low-fidelity data has large deviations/distribution shifts
from high-fidelity data. This is illustrated by the visualization of low-fidelity data generated from
the two: the downsampled one contains more fine-grained details than the solver-based one.

To address this, we study reconstructing high-060 fidelity CFD data from solver-generated low-061 fidelity data with initial coarser discretiza-062 tion grids (Sarkar et al., 2023; Ogoke et al., 063 2024). Problem: Our experiments reveal 064 that state-of-the-art models struggle to recover fine-grained high-fidelity details given solver-065 generated low-fidelity inputs, due to the large 066 distribution shifts. In response to this chal-067 lenge, we present a novel diffusion model, 068 PG-Diff, which features an Importance Weight 069 strategy during training as self-guidance to locate fine-grained high-fidelity details, and a 071 training-free Residual Correction module during inference for injecting physical inductive 073 bias. The two moduli jointly guide the dif-074 fusion model toward higher-quality reconstruc-075 tions. Specifically, the Importance Weight mod-



Figure 1: Comparison between downsampled (**black** line) and solver-generated (**red** line) flow fields. Solver-generated low-fidelity data retain less information, especially for fined-grained high-fidelity details.

ule assigns importance scores to different components in the flow field through Discrete Wavelet 076 Transformation (DWT) (Daubechies, 1992a), which are integrated into the loss function for guiding 077 the diffusion model to better reconstruct detailed and accurate structures. The Residual Correction module projects the reconstructed samples onto the solution subspace of the data governing equa-079 tions, ensuring that the generated outputs also adhere to physical constraints such as Navier-Stokes Equations (Navier, 1823). We achieve this by applying gradient descent of the residuals of the gov-081 erning equations at certain diffusion steps, to refine the generated high-fidelity data. We also explore 082 different scheduling strategies on which diffusion steps to apply such correction. Our findings sug-083 gest that applying residual corrections at both the beginning and the end of diffusion steps strikes 084 the ideal balance between reconstruction L2 error and the physical coherence measured by PDE 085 residual.

Our key contributions are summarized as follows:

- We study a novel problem on reconstructing high-fidelity flow fields with solver-generated low-fidelity data, benefiting real-world applications. Our experiments reveal that state-of-the-art reconstruction models fail to generate high-quality outputs due to loss of fine-grained detail in low-fidelity data.
- We propose PG-Diff, a novel diffusion model for reconstructing high-quality outputs through the guidance of an *Importance Weight* strategy during training as self-guidance and a training-free *Residual Correction* method during inference as physical inductive bias.
- We present empirical evidence of PG-Diff's state-of-the-art performance in a variety of 2D turbulent flow over 4 datasets. It yields a significant improvement in terms of predictive accuracy physical consistency, and perceptual quality.
- 2 PRELIMINARIES AND RELATED WORK
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We consider a machine learning model $f_{\theta} : \mathcal{X} \to \mathcal{Y}$ with parameters θ , which transforms a data sample from low-fidelity domain $x \in \mathcal{X} \in \mathbb{R}^{m \times m}$ to high-fidelity domain $y \in \mathcal{Y} \in \mathbb{R}^{n \times n}(m < n)$. The distributions of the training and test sets for low-fidelity data are denoted by $p_{\mathcal{X}}^{\text{train}}$ and $p_{\mathcal{X}}^{\text{test}}$, respectively, and for high-fidelity data as $p_{\mathcal{Y}}^{\text{train}}$ and $p_{\mathcal{Y}}^{\text{test}}$. The training and testing distribution for low and high-fidelity data are not necessarily identical. The objective is to develop f_{θ} such that it can effectively map samples from $\mathcal{X}^{\text{test}}$ to their corresponding high-fidelity counterparts $\mathcal{Y}^{\text{test}1}$. As diffusion model operates on the same input and output grid, we upsample the low-fidelity data uniformly to the same resolution as the target.

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2.1 AI FOR COMPUTATIONAL FLUID DYNAMICS (CFD)

114 Recent advances in machine learning have led to various learning-based surrogate models for accel-115 erating scientific discoveries (Sanchez-Gonzalez et al., 2020; Li et al., 2020; 2021). In the field of 116 high-fidelity CFD reconstruction, researchers have developed powerful models rooted from image 117 super-resolution domain in computer vision. Specifically, GANs (Ledig et al., 2016; Wang et al., 2019) and normalizing flows (Lugmayr et al., 2020) have achieved impressive results for image re-118 construction. Later on, diffusion models (Saharia et al., 2021) have challenged the long-standing 119 dominance of GANs. Various physical inductive biases (Raissi et al., 2019; Bai et al., 2020; Chen 120 et al., 2021) have been injected into these methods for CFD reconstructions, which enhances the 121 robustness and accuracy of the reconstruction task. For example, (Erichson et al., 2020; Pradhan 122 & Duraisamy, 2021) developed super-resolution models based on Multi-Layer Perceptrons (MLPs). 123 Fukami et al. (2019) introduced a CNN-based hybrid Downsampled Skip-Connection Multi-Scale 124 (DSC/MS) model, while Fukami et al. (2021) adapted existing CNN models for use with moving 125 sensors. Li & McComb (2022) further advanced the field by proposing physics-informed GANs 126 for super-resolving multiphase fluid simulations. Additionally, Fu et al. (2023) proposed a porposal 127 and refinement network to address the issue with limited high-fidelity data. Ren et al. (2023); Jiang 128 et al. (2020) also leveraged CNN for spatial-temporal super resolution. These models rely on lowand high-fidelity data pairs during training and can only capture specific relationships between these 129 pairs. As a result, when the test data deviates significantly from the training data, their performance 130 degrades. To overcome this limitation, Gao et al. (2021) utilizes physical properties of the fluid 131 such as conservation laws and boundary conditions for super resolution. Shu et al. (2023) lever-132 aged diffusion model, which is trained exclusively on high-fidelity data and enables reconstruction 133 from any type of low-fidelity input. However, these methods assume the low-fidelity data is arti-134 ficially downsampled from high-fidelity sources, which limits their performance during inference 135 when reconstructing from solver-generated low-fidelity data in reality.

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2.2 DENOISING DIFFUSION PROBABILISTIC MODEL

139 Diffusion models have become a prominent class of deep generative models, demonstrating state-140 of-the-art performance across various domains such as image generation (Meng et al., 2022; Saharia 141 et al., 2021; Ho et al., 2020), video synthesis (Yang & Hong, 2022; Yu et al., 2022), 3D shape 142 creation (Zhou et al., 2021; Zeng et al., 2022), and applications in scientific fields (Shu et al., 2023; 143 Yang & Sommer, 2023; Qiu et al., 2024). Denoising Diffusion Probabilistic Models (DDPMs) are 144 grounded in a stochastic diffusion process, akin to those found in thermodynamics. It contains a 145 forward and reverse process, where a data sample is gradually corrupted with noise via a Markov chain, and a neural network is trained to reverse this, progressively removing the noise. To generate 146 new samples, a fully noisy input is denoised by the model step by step. 147

148 Formally, in the forward diffusion process, the input data x_0 is gradually corrupted with Gaussian 149 noises as $q(x_t|x_{t-1}) = \mathcal{N}(x_t; \sqrt{1-\beta_t x_{t-1}}, \beta_t I)(t = 1, 2, \cdots T)$, where β_1, \ldots, β_T are fixed variance schedule that control the amount of noise introduced at each step, T is the number of 150 total diffusion steps and $q(x_t|x_{t-1})$ is the Markov transition probability. Let $\bar{\alpha}_t := \prod_{s=1}^t 1 - \beta_s$, 151 we have $x_t = \sqrt{\overline{\alpha}_t} x_0 + \sqrt{1 - \overline{\alpha}_t} \epsilon_t$, which describes how to generate noisy states x_t from input 152 x_0 . The reverse process is defined as $p_{\theta}(x_{t-1}|x_t) = \mathcal{N}(x_{t-1}; \mu_{\theta}(x_t, t), \Sigma_{\theta}(x_t, t))$, where μ_{θ} and 153 Σ_{θ} are learned by neural networks parameterized by θ , and $p_{\theta}(x_{t-1}|x_t)$ is the denoising transition 154 probability, undoing the transformation in the forward process. 155

Training DDPM involves minimizing a variational bound on the negative log-likelihood of $q(x_0)$. Ho et al. (2020) showed that it can be simplified as predicting the noise at each step:

$$L^{\text{simple}} = \mathbb{E}_{t,x_0,\epsilon} \left[\|\epsilon_t - \epsilon_\theta (\sqrt{\bar{\alpha}_t} x_0 + \sqrt{1 - \bar{\alpha}_t} \epsilon_t, t) \|^2 \right], \tag{1}$$

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¹For direct mapping models, the training takes low- and high-fidelity pairs as inputs. For diffusion models, the training only requires high-fidelity data, and low-fidelity data is used as input during testing.

where $\epsilon_{\theta}(\cdot)$ is the predicted noise from DDPM's denoiser neural network. The denoiser takes noised sample $x_t = \sqrt{\bar{\alpha}_t} x_0 + \sqrt{1 - \bar{\alpha}_t} \epsilon_t$ and diffusion timestamp t as input to predict noise at each timestamp. $\epsilon_t \sim \mathcal{N}(\mathbf{0}, \mathbf{I})$ is the standard Gaussian noise sampled at time t. In this paper, we implement the backbone diffusion model proposed by Ho et al. (2020) and apply accelerated sampling techniques introduced by Song et al. (2020).



Figure 2: Training and inference pipeline of PG-Diff. Training with high-fidelity data only, guided by importance weight strategy to locate fine-grained high-fidelity details. During inference, low-fidelity data is used for reconstruction, and residual correction is applied at intermediate diffusion steps to improve physical coherence. We follow Ho et al. (2020) and use U-net as our denoiser.

3 Method:PG-Diff

We present a novel diffusion framework PG-Diff for reconstructing high-fidelity CFD from solver-generated low-fidelity input, which has larger distribution shifts and more information loss compared to artificially downsampled low-fidelity inputs. PG-Diff features an *Importance Weight* strategy that scores different components in the flow fields through the loss function as self-guidance, forcing the model to recover more fine-grained high-fidelity details during training. In addition, a training-free *Residual Correction* module applies physics-informed correction during inference, ensuring physical coherence in reconstructed samples. The two moduli jointly guide the model towards highquality reconstruction from wide-range of low-fidelity inputs. The overall framework is depicted in Figure 2. We now introduce each component in detail.

Model Setup. PG-Diff follows the guided data synthesis setting for CFD reconstruction as in (Shu et al., 2023): we train the model via recovering high-fidelity sources only, and condition on low-fidelity inputs as intermediate diffusion step during inference. This allows the model to 1.) exert control over the data generation process (reverse diffusion) during inference. Instead of starting from random noises, starting from low-fidelity inputs as intermediate diffusion step sintermediate diffusion steps; and 2.) reconstruct from any form of low-fidelity data, as the training does not depend on low- and high-fidelity pairs.

Formally, during the training forward process, we obtain intermediate diffusion states via the following, where $y \sim p_{\mathcal{Y}}^{\text{train}} := x_0$ is the high-fidelity CFD that we want to recover. Here ϵ_t is sampled from a standard Gaussian distribution. $\bar{\alpha}_t := \prod_{s=1}^t 1 - \beta_s$ as introduced in Sec 2.2.

$$x_t = \sqrt{\overline{\alpha}_t} x_0 + \sqrt{1 - \overline{\alpha}_t} \epsilon_t = \sqrt{\overline{\alpha}_t} y + \sqrt{1 - \overline{\alpha}_t} \epsilon_t, \quad t = 1, 2, \cdots T.$$
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216 During the inference stage, we first obtain the intermediate diffusion state $x_{t_{guide}}(0 < t_{guide} < T)$ by 217 sequentially adding noises to the conditioned low-fidelity data $x \sim p_{\chi}^{\text{test}}$ as 218

$$x_{t_{\text{guide}}} = \sqrt{\overline{\alpha}_{t_{\text{guide}}}} x + \sqrt{1 - \overline{\alpha}_{t_{\text{guide}}}} \epsilon_{t_{\text{guide}}}.$$
(3)

We then use $x_{t_{euide}}$ as the starting point for the reverse process to reconstruct the high-fidelity sources. Equivalently, this means the reverse diffusion starts from t_{guide} instead of T. The reverse diffusion process progressively produces a refined high-fidelity reconstruction that aligns with the low-fidelity conditioning data, using the following DDIM (Song et al., 2020) sampling formula for acceleration.

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 $x_{t-1} = \sqrt{\bar{\alpha}_{t-1}} x_0^t + \sqrt{1 - \bar{\alpha}_{t-1}} - \sigma_t^2 \epsilon_\theta(x_t, t) + \sigma_t^2 \epsilon_t, \quad 0 < t < t_{\text{guide}}$ (4)

where $x_0^t = \frac{x_t - \sqrt{1 - \bar{\alpha}_t} \epsilon_{\theta}(x_t, t)}{\sqrt{\bar{\alpha}_t}}$ is the predicted reconstruction at each reverse diffusion step t. We 227 228 use U-net as in Ho et al. (2020) to serve as our denosier, detailed in Appendix C.2.

3.1 IMPORTANCE WEIGHT DURING TRAINING

Existing diffusion models can fail to capture high-fidelity fine-grained details when conditioned on 232 solver-generated low-fidelity data. We therefore introduce an importance weighting mechanism dur-233 ing training as self-guidance to address this limitation. Specifically, we transform the high-fidelity 234 data into the wavelet domain using the DWT and assign importance scores to different components 235 of fluid fields within the diffusion loss function. DWT has great abilities to locate information 236 both in spatial and frequency domains (Daubechies, 1992b; Akansu & Haddad, 1992), thus allow-237 ing the model to capture fine-grained structures more effectively. Formally, we decompose fluid 238 fields $y \in \mathcal{R}^{n \times n}$ into frequency subdomains, and compute the sum of squares of the high-frequency 239 modes - namely HL (high-low), LH (low-high) and HH (high-high) subdomains as below, where $HL, LH, HH, F \in \mathbb{R}^{\frac{n}{2} \times \frac{n}{2}}$. HL captures horizontal high-frequency signals, while LH captures ver-240 241 tical high-frequency signals. HH captures high-frequency signals in both directions, corresponding to diagonal details such as intersections. Details about DWT are described in Appendix C.1. 242

$$F = HL^2 + LH^2 + HH^2.$$
 (5)

To compute importance scores, we then uniformly upsample F to $\hat{F} \in \mathbb{R}^{n \times n}$ and linearly map 245 $\hat{F}_{i,j}$ to an importance weight $a_{i,j}$ as below, where α, β are the minimum and maximum importance 246 247 weight value respectively, $Q_{\theta}(\hat{F}) \in \mathbb{R}$ is the θ quantile of all \hat{F} values.

$$a_{i,j} = \begin{cases} \alpha + (\beta - \alpha) \frac{\hat{F}_{i,j} - Q_{\theta}(\hat{F})}{\max \hat{F} - Q_{\theta}(\hat{F})} & \text{if } \hat{F}_{i,j} > Q_{\theta}(\hat{F}) \\ 1 & \text{otherwise} \end{cases}$$
(6)

If $\hat{F}_{i,j}$ exceeds the θ quantile of all \hat{F} values, the corresponding component is considered highfrequency, and will be assigned with weight greater than 1. Finally, the diffusion loss function in 253 Eqn 1 is updated to incorporate the importance weighting as follows: 254

$$L^{\text{IW}} = \mathbb{E}_{t,y,\epsilon} \left[a \odot \left\| \epsilon_t - \epsilon_\theta \left(\sqrt{\bar{\alpha}_t} \, y + \sqrt{1 - \bar{\alpha}_t} \, \epsilon_t, \, t \right) \right\|^2 \right],\tag{7}$$

where ϵ_{θ} represents the predicted noise by the denoiser, ϵ_t is the ground truth noise at time t, and \odot 258 denotes element-wise multiplication.

259 Importance Weight Design Choice. Our importance weight strategy is incorporated exclusively 260 during training, ensuring that the model focuses on regions with high-frequency details. The DWT-261 based calculation avoids the large computational complexity often associated with attention mech-262 anisms. It efficiently emphasizes important features by leveraging the intrinsic properties of the 263 wavelet transform, resulting in a more targeted learning process. 264

265 3.2 **Residual Correction During Inference** 266

267 In addition, we introduce a physics-informed residual correction module to enhance the physical 268 coherence of the reconstructed data during inference. In the reverse diffusion process, we apply such correction to refine the reconstructed high-fidelity data at certain diffusion steps t determined by a 269 scheduling policy. Since we use DDIM (Song et al., 2020) for acceleration, we represent the sampled 270 diffusion states as x^{τ} , and its corresponding reconstruction as $x_0^{\tau}(\tau < t_{guide})$ in the following. Note 271 that we do not apply correction on the noisy state $x^{ au}(au < t_{ ext{guide}})$ directly. The correction is to 272 perform gradient descent based on the residuals of the governing PDEs, detailed in Appendix D.1. 273 The key parameters in this process are the residual correction schedule policy (the specific diffusion 274 steps at which the correction is applied), the number of gradient descent steps N, and the step size η , which we study their impacts in Exp 4.5. Our findings suggest that applying residual corrections 275 at both the beginning and the end of diffusion steps strikes the ideal balance between reconstruction 276 L2 error and the physical coherence. We utilize the Adam algorithm Kingma & Ba (2015) for the 277 gradient descent and summarize the physics-guided inference procedure using DDIM (Song et al., 278 2020) in Algorithm 1. 279

One existing work proposes a conditional diffusion model (Shu et al., 2023), where residual at every 280 training diffusion step is concatenated with the current state to generate the diffusion state at next 281 timestamp, serving as the condition. One key advantage of PG-Diff is that it is training-free, allow-282 ing the correction process to be applied without additional learning to the original diffusion model. 283 Also, our approach offers the flexibility to adjust the residual correction schedule dynamically. This 284 adaptability enables an optimal balance between predictive accuracy and physical consistency of 285 the reconstructed sample. We empirically show our method achieves better performance in Sec 4.3. Additionally, diffusion model guided generation has been widely studied in Chung et al. (2023); 287 Huang et al. (2024); Zhu et al. (2023); Shysheya et al. (2024), our Residual Correction differs by ap-288 plying guidance to denoised samples instead of noisy samples. Our physical guidance also involves 289 multiple steps of Adam gradient descent at selected backward diffusion steps, in contrast to a single 290 step of gradient descent applied at every backward diffusion step in other works.

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Algorithm 1 Physics-informed Training-free Correction with DDIM during Inference.

Require: $x \in \mathcal{X}^{\text{test}}$ (guide), t_{guide} ($t_{\text{guide}} < T$), $\tau = \{\tau_0, \tau_1, ..., \tau_K\}$ (sampled diffusion timestamp sequence, where $\tau_0 = 0, \dots \tau_K = t_{\text{guide}}$. ϵ_{θ} (a trained DDPM model), $\mathcal{R}(\cdot)$ (residual of the governing PDE), N (number of gradient descent steps), η (step size).

296 1: $\epsilon_{\tau_K} \sim \mathcal{N}(0, I)$ 2: $x_{\tau_K} = \sqrt{\bar{\alpha}_{\tau_K}} x + \sqrt{1 - \bar{\alpha}_{\tau_K}} \epsilon_{\tau_K}$ 3: **for** $i = K, K - 1, \dots, 1, 0$ **do** # Adding noises to low-fidelity input via forward diffusion # Reverse diffusion from τ_K ($\tau_K = t_{guide}$) 297 298 299 $x_0^{\tau_i} = \frac{x_{\tau_i} - \sqrt{1 - \bar{\alpha}_{\tau_i}} \epsilon_{\theta}(x_{\tau_i}, \tau_i)}{\sqrt{\bar{\alpha}_{\tau_i}}}$ if correction is performed at time τ_{i-1} then 300 4: # Predicted reconstruction (x_0) at τ_i 301 5: 302 6: # Residual-Based Correction 303 $\begin{array}{l} \textbf{repeat} \\ x_0^{\tau_i} = x_0^{\tau_i} - \eta \cdot \operatorname{Adam}(\nabla \mathcal{R}(x_0^{\tau_i})) \\ \textbf{until} \; N \; \text{times} \end{array}$ 7: 304 8: 305 9: end if 10: 306 $x_{\tau_{i-1}} = \sqrt{\bar{\alpha}_{\tau_{i-1}}} x_0^{\tau_i} + \sqrt{1 - \bar{\alpha}_{\tau_{i-1}} - \sigma_{\tau_i}^2} \epsilon_\theta(x_{\tau_i}, \tau_i) + \sigma_{\tau_i}^2 \epsilon_{\tau_i}. \qquad \# x_{\tau_{i-1}} \sim p_\theta(x_{\tau_{i-1}} | x_{\tau_i})$ 11: 307 12: **end for** 308 13: **return** $y = x_0$ 309 310 311 312 313 4 EXPERIMENTS 314 315 4.1 DATASET 316

We generated four 2D turbulent flow datasets with different characteristics: 1.) *Taylor Green Vortex*,
 featuring how vortices diminish in turbulent flows, where large-scale vortices gradually break down
 into smaller turbulent structures; 2.) *Decaying Turbulence*, describing turbulence that evolves naturally without external forces. As time progresses, the turbulence weakens due to viscous effects; 3.)
 Kolmogorov Flow, which portrays turbulence influenced by a sinusoidal external force combined
 with a drag component; 4.) *McWilliams Flow* (Mcwilliams, 1984), which describes the behavior
 of isolated vortices in turbulent conditions. It is the most challenging one, demanding accurate modeling of inverse energy transfer and multi-scale vortex interactions.

The four datasets are generated by the incompressible Navier-Stokes equations following (Li et al., 2021; Kochkov et al., 2021)

$$\begin{split} \frac{\partial \omega(\boldsymbol{x},t)}{\partial t} + \boldsymbol{u}(\boldsymbol{x},t) \cdot \nabla \omega(\boldsymbol{x},t) &= \frac{1}{Re} \nabla^2 \omega(\boldsymbol{x},t) + f(\boldsymbol{x}), \\ \nabla \cdot \boldsymbol{u}(\boldsymbol{x},t) &= 0, \quad \omega(\boldsymbol{x},0) = \omega_0(\boldsymbol{x}), \end{split}$$

330 where ω represents the vorticity, u denotes the velocity field, Re is the Reynolds number, and 331 f(x) is an external forcing term. ω_0 represents the initial vorticity distribution. The PDE is nu-332 merically solved by pseudo-spectral solver (Orszag, 1972) on equispaced discretization grids. The 333 high-fidelity data are generated with 2048×2048 discretization grid and then uniformly downsam-334 pled to 256×256 , while those on the lower-resolution grids are considered low-fidelity. For each 335 dataset, we use 80% of the trajectories for training. 10% for validation, and 10% for testing. More details can be found in Appendix D. Note that PG-Diff also utilizes the PDE for residual correction 336 discussed in Sec 3.2. 337

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4.2 EXPERIMENT SETTINGS

341Task Setup and Baselines. We evaluate two reconstruction settings with different low-fidelity res-342olutions: $64 \times 64 \rightarrow 256 \times 256$ ($4 \times$ upsampling) and $32 \times 32 \rightarrow 256 \times 256$ ($8 \times$ upsampling).343To benchmark our approach, we compared against both direct mapping models and diffusion mod-344els: bicubic interpolation (Gonzalez & Woods, 2007), a CNN-based model (Fukami et al., 2019),345a GAN-based model (Li & McComb, 2022), the vanilla diffusion model (Diff) and its conditional346variant (Cond Diff) from Shu et al. (2023). In addition, we perform two ablation studies, namely PG-347respectively. More implementation details are provided in Appendix C.2.

348 Evaluation Metrics. We assess the reconstructed flow fields using three key metrics. We first 349 use two standard metrics suggested by Shu et al. (2023): L2 norm for measuring the pointwise 350 error between prediction and ground truth; unnormalized residuals of the governing equation (Res.) 351 for assessing adherence to the underlying physics. In addition, we conduct a novel multi-scale 352 evaluation using DWT: we transform the predicted and ground truth flow fields into wavelet space 353 and decompose them into four subdomains: LL (low-low), LH (low-high), HL (high-low), and HH 354 (high-high). The LL subdomain captures large-scale, low-frequency information, while LH, HL, 355 and HH encompass higher-frequency details like turbulent structures. By calculating the L2 norm in each subdomain, we gain a comprehensive understanding of the model's performance across 356 different scales, ensuring accurate reconstruction of both global flow features and fine-scale details. 357

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4.3 RECONSTRUCTION RESULTS

Table 1 reports the mean and standard deviation of L2 and PDE residual across datasets and models. 361 For both upsampling scales $(32 \times 32 \rightarrow 256 \times 256 \text{ and } 64 \times 64 \rightarrow 256 \times 256)$, PG-Diff consistently 362 outperforms baselines, showing its effectiveness. Notably, it achieves 3.5% to 7.7% performance 363 gain against baselines in the $4 \times$ upsampling setting. The lower L2 achieved by PG-Diff indicates 364 that PG-Diffeffectively captures the essential features and dynamics of the turbulent flows, while the significantly reduced PDE residuals demonstrate that PG-Diff's predictions adhere more closely 366 to the underlying physical laws governing fluid dynamics. Reconstructions from direct mapping 367 models (GAN, CNN) exhibit significant deviations in physical coherence. For complex datasets 368 dominated by fine-grained details, such as Kolmogorov Flow and McWilliams Flow, PG-Diff consistently outperforms baselines by a big margin, showing better reconstruction accuracy and physical 369 coherence. Through ablation studies, we demonstrate that both Importance Weight and Residual 370 Correction contribute significantly in improving model's performance, underscoring the effective-371 ness of our design choices in capturing the complex behaviors of turbulent flows. 372

Multi-Scale Evaluation. In addition, we assess the model's ability to capture flow structures at different scales using DWT. As shown in Figure 12, PG-Diff demonstrates superior performance in the LL, LH, and HL subdomains, achieving the best or near-best results among all methods. Excelling in the LL subdomain indicates a strong capability in capturing large-scale, low-frequency components of the turbulent flows. The superior performance in the LH and HL subdomains suggests the effectiveness of PG-Diff in capturing small-scale vortices and transitions between scales. While

		Bicubic*	CNN	GAN	Diff	Cond Diff	PG-Diff	PG-Diff w/o Cor	PG-Diff w/o IW
nə	L2	6.08	3.10±0.07	3.36±0.08	3.28±0.01	4.50±0.04	$\frac{3.18\pm0.01}{5.212\pm0.01}$	3.20±1e-4	3.22±0.01
ire X	Res.	2.18e4	$1.40e5 \pm 3.6e4$	$4.54e4 \pm 1.3e4$	$4.1/e4 \pm 3.3e3$	1.36e3±42	$5.31e3\pm40$	$1.44e4 \pm 1.4e4$	6.95e3±569
Taylor G Vorte	L2 Res.	3.04 <u>1.05e4</u>	2.99±0.06 1.37e5±3.6e4	2.83±0.3 4.18e4±1.1e4	$\frac{1.68\pm2e-3}{3.06e4\pm4.3e3}$	3.57±0.02 1.47e5±1.9e4	1.55±9e-3 4.39e3±3	1.57±0.02 6.67e4±8.4e3	1.59±6e-3 6.09e3±9
ng Ing	L2 Res.	3.62 3.41e3	$\frac{1.81\pm0.1}{3.05e3\pm500}$	2.09±0.02 7.63e2±111	1.84±1e-3 9.97e3±460	2.02±8e-4 3.49e4±1.7e4	1.78±1e-3 1.65e2±11	1.78±2e-3 9.86e2±165	1.79±3e-3 2.84e2±22
Decayir Turbulen	L2 Res.	1.74 1.31e3	1.66±0.1 3.16e3±781	2.06 ± 0.05 $6.65 e 2 \pm 59$	$\frac{0.85 \pm 6e{-}4}{4.33e3 \pm 164}$	1.34±7e-4 1.52e4±161	0.82±1e-3 85±0.9	0.83±6e-3 2.76e3±1.5e3	0.83±4e-3 1.87e2±37
nov	L2 Res.	4.71 2.63e3	3.02±8e-3 3.52e3±125	3.14±0.04 3.30e2±87	$\frac{3.09\pm1e-3}{1.80e2\pm22}$	$3.13 \pm 1e-3$ $\underline{80 \pm 2}$	2.78±8e-3 40.12±0.3	2.82±1e-3 6.68e2±10	2.93±5e-4 1.40e2±83
Kolmoge Flow	L2 Res.	3.22 1.86e4	2.22±0.01 7.30e2±49	3.02±0.04 3.03e2±92	$\frac{1.79\pm1e-3}{3.09e2\pm29}$	$\frac{1.79 \pm 1e-3}{1.65e2 \pm 7}$	1.66±4e-3 39±0.06	1.69±8e-4 9.27e2±1	1.73±1e-3 3.33e2±145
ams	L2 Res.	3.19 3.35e2	2.00±0.07 6.50e2±180	$\frac{2.03\pm0.05}{2.86e2\pm51}$	2.23±4e-3 16±0.9	2.24±5e-4 12±1	$\frac{2.04{\pm}2\text{e-}4}{\textbf{5.5}{\pm}\textbf{4\text{e-}3}}$	2.06±3e-5 66±1	2.16±3e-4 7.8±0.2
McWilliu Flow	L2 Res.	2.17 2.17e2	1.75±0.1 3.64e2±127	1.96±0.08 2.42e2±35	$\frac{1.29 \pm 3e \cdot 4}{\underline{21 \pm 4}}$	1.30±4e-4 31±4	1.24±9e-5 6.36±2e-3	1.27±1e-4 88±0.7	1.30±2e-4 12±0.06

Table 1: Quantitative performance comparison over four datasets on L2 and Res. Results are repeated three times. Metrics are reported for both $32 \times 32 \rightarrow 256 \times 256$ (grey) and $64 \times 64 \rightarrow 256 \times 256$ tasks. Bold values indicate the best performance and underlined values represent the second-best.

* Bicubic interpolation is a deterministic algorithm, which has zero standard deviation.

> direct mapping models (CNN, GAN) exhibit better performance in the HH subdomain, PG-Diff's results are close to those best-performing methods. The balanced performance across all subdomains demonstrates that PG-Diff can reconstruct both large-scale structures and fine-grained details essential for turbulent flows. Additional details on other datasets are presented in the Appendix E.2.

Runtime Comparison. We report the runtime comparison of different methods in Appendix E.8. PG-Diff only increases small inference time while achieving superior performance improvement against baselines. The total runtime of using the numerical solvers to generate low-fidelity data, and then reconstructing with PG-Diff is considerably faster than the time required to produce samples with similar error, demonstrating the effectiveness of PG-Diff.

Sensitivity Analysis. We study the effects of three key hyperparameters in calculating the impor-tance weight in Sec 3.1: the maximum importance weight β , minimum importance weight α , and the threshold parameter θ . The results in Appendix E.5 show that increasing β reduces both L2 and PDE residuals, indicating a broader range of importance weights is beneficial. Smaller α values improve performance, while setting α to 1 is suboptimal. This can be understood as it decreases the difference between high-frequency and low-frequency regions, as the weight for the latter one is set to 1. Additionally, a larger θ (0.7 or 0.8) helps the model focus on important details.

- 4.4 VISUALIZATIONS

We visualize the reconstructed high-fidelity data conditioned on low-fidelity inputs in Figure 3. PG-Diff consistently captures more fine-grained details, resulting in a closer resemblance to the high-fidelity ground truth compared to other methods. This is particularly evident in regions with complex vortical structures and turbulent features, where competing models often smooth out finer details. An example is the reconstructions from CNN and GAN on the McWilliams FLow dataset. While direct mapping methods can achieve a low L2 norm, their reconstructed samples often merge small fine-grained regions into one large blurred region, leading to information loss and significant divergence from the original high-fidelity data. By comparison, PG-Diff is able to recover the sharp edges, turbulence, and subtle variations in the flow field. We additionally show the strong reconstruction ability of our model variants compared to baselines in Appendix E.3. We also use the Learned Perceptual Image Patch Similarity (LPIPS) score (Zhang et al., 2018) to measure the perceptual quality of reconstructed samples in Appendix E.4.



Figure 3: Visualization of reconstructed high-fidelity data across datasets. We highlight fine-grained details where PG-Diff achieves better performance than baseline diffusion models.

4.5 PHYSICAL GUIDANCE

We conduct a systematic study on how to
schedule the residual correction during inference towards the best performance.

457 Scheduling Policy. We first explore what 458 would be the optimal schedule policy 459 by comparing against: 1.) Uniform N: 460 Distributing N residual correction steps 461 evenly across diffusion steps; 2.) Start M, 462 End N Placing M consecutive correction 463 steps at the start and N at the end of the diffusion process; 3.) Start N, Space S: 464 Placing N correction steps at the start with 465 a spacing of S; 4.) End N, Space S: Placing 466 N correction steps at the end with a spac-467 ing of S. Results in Table 2 suggest that 468 applying correction steps at the beginning 469 achieves the lowest L2, while applying at 470 the end has the lowest PDE residuals. To 471 balance between L2 and PDE residual, we

Schedule	32×32 -	→ 256×256	$64{\times}64{\rightarrow}256{\times}256$		
	L2	Res.	L2	Res.	
Uniform 4	2.7910	26.32	1.6645	23.50	
Start 3 End 1	2.7906	30.00	1.6617	32.51	
Start 2 End 2	2.7897	40.12	1.6609	39.97	
Start 1 End 3	2.7896	59.79	1.6617	59.39	
Start 4 Space 1	2.7889	101.60	1.6587	105.84	
Start 4 Space 2	2.7888	98.32	1.6590	102.14	
Start 4 Space 3	2.7887	95.36	1.6594	95.08	
End 4 Space 1	2.7930	19.30	1.6668	21.97	
End 4 Space 2	2.7932	17.55	1.6747	20.96	
End 4 Space 3	2.7924	19.26	1.6730	23.57	

Table 2: Scheduling policy comparison on *Kolmogorov Flow*. We emphasize that applying residual correction at the end leads to a reduced PDE residual, while placing it at the beginning achieves a lower L2 loss.

adopt the *Start N End N* schedule, which places N correction steps at both the beginning and the end of the diffusion process. We next study the optimal number of N.

Number of Correction Steps. We vary the number of N in the *Start N, End N* policy as shown in Figure 4. We observe that increasing N leads to enhanced physical coherence measured by PDE residuals. However, L2 does not continuously decrease with larger N. It reaches a minimum when N = 2. This suggests that while more correction steps improve models' adherence to physical laws, an excessive number may interfere with the model's ability to accurately capture the intricate details of the turbulent flow. Therefore, we use *Start 2, End 2* as the optimal balance in our method.

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4.6 MODEL GENERALIZATION

We observe that PG-Diff generalizes well even beyond its training distributions. Specifically, we conduct evaluations over three generalization settings: time discretization in numerical solver, spatial domain size, and Reynolds number as shown in Table 3. We train PG-Diff on the original *Kolmogorov Flow* dataset configurations, which has solver integration timestep as dt = 1/32, spa-



Figure 4: Varying number of N using the *Start N, End N* correction schedule on the *Kolmogorov Flow* dataset. Showing both L2 and Residual across upsampling settings.

tial domain size ranging from $2\pi \times 2\pi$, and Reynolds number as Re = 1000. We then directly test these new configurations on the trained model, without any additional retraining or fine-tuning. We compare the performance against models directly trained one each new configuration.

501 The results reveal that the per-502 tained PG-Diff performs comparably to trained ones directly on each new 503 configuration. This underscores our 504 model's strong generalization capa-505 bilities across different flow condi-506 tions. Such generalization ability can 507 due to the fact that both our impor-508 tance weight mechanism and resid-509 ual correction modules are training-510 free. They enable our model to lo-511 cate fine-grained high-fidelity details 512 and adhere to physical laws indepen-513 dent of training data, showing their strong superiority. A similar trend is 514 515 also observed in $4 \times$ upsampling experiments, shown in Appendix E.7. 516

5 CONCLUSION

519
520 We study a novel problem of reconstructing high-fidelity CFD from
521 solver-generated low-fidelity inputs
523 in a practical setting across scientific
524 domains. We present a novel diffu-

Variation Model L2 Res. **Time Discretization Variations** 34.29 dt = 1/40Trained on Original Data 2.8144 Trained on dt = 1/40 Data 2.8136 34.52 2.7761 37.11 dt = 1/50Trained on Original Data 2.7795 Trained on dt = 1/50 Data 30.60 **Spatial Domain Size Variations** $1\pi \times 1\pi$ Trained on Original Data 1.9045 262.42 Trained on $1\pi \times 1\pi$ Data 1.7672 190.99 2.3573 $1.5\pi \times 1.5\pi$ Trained on Original Data 80.66 2.3028 88.88 Trained on $1.5\pi \times 1.5\pi$ Data **Reynolds Number Variations** Re = 500Trained on Original Data 2.3694 24.11 Trained on Re = 500 Data 2.3542 26.96 Re = 2000Trained on Original Data 3.2739 21.21 Trained on Re = 2000 Data 3.2914 22.48

Table 3: Generalization results on *Kolmogorov Flow* dataset with $32 \times 32 \rightarrow 256 \times 256$ setting.

sion model PG-Diff, which achieves high-quality reconstruction guided by the proposed *Importance Weight* and *Residual Correction* modulus jointly, yielding state-of-the-art performance across four 2D turbulent flow datasets.

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AL	nitations
B B	oader Impacts
СМ	odel Details
C	Discrete Wavelet Transformation
C	2 Implementation Details
D D	taset
D	1 Residual Calculation
ΕA	ditional Experimental Results
E.	DWT visualization
E.	2 Multi-Scale Evaluation
E.	Reconstruction Visualization
E.	LPIPS Scores
	5 Importance Weight Sensitivity Analysis
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E. E.	6 Comparison of Residual Correction and Post Processing
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A LIMITATIONS

Our current research on super-resolution in computational fluid dynamics (CFD) focuses exclusively
 on vorticity data. However, future work can broaden its scope to include the reconstruction of high fidelity velocity and pressure fields. These additional physical quantities are critical for capturing a
 more comprehensive representation of fluid dynamics.

B BROADER IMPACTS

By enhancing the accuracy of CFD simulations, this approach can significantly reduce the computational costs associated with high-resolution simulations, which are often prohibitively expensive in terms of both time and resources. This has implications for a wide range of industries, including aerospace, automotive design, and environmental engineering, where high-fidelity simulations are essential for optimizing performance, safety, and sustainability. Additionally, by improving the quality of low-resolution CFD data, our work could enable more accessible, efficient research and development processes, allowing smaller organizations and research teams to leverage advanced simulation capabilities.

⁸¹⁰ C MODEL DETAILS

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812 C.1 DISCRETE WAVELET TRANSFORMATION

The Discrete Wavelet Transform (DWT) decomposes an image $I \in \mathbb{R}^{n \times n}$ into different frequency components by successively applying low-pass and high-pass filters. Given an image I, the DWT first applies a low-pass filter g[n] and a high-pass filter h[n] along each row, producing two subbands: $I_{\text{low}}[i, j] = \sum_k I[i, k] \cdot g[2j - k]$ and $I_{\text{high}}[i, j] = \sum_k I[i, k] \cdot h[2j - k]$, each downsampled by a factor of 2 along the columns, resulting in matrices of shape $n \times \frac{n}{2}$.

Next, the same filters are applied to each column of I_{low} and I_{high} , generating four sub-bands:

$$LL[i, j] = \sum_{k} I_{\text{low}}[k, j] \cdot g[2i - k], \quad LH[i, j] = \sum_{k} I_{\text{low}}[k, j] \cdot h[2i - k],$$

$$HL[i,j] = \sum_{k} I_{\text{high}}[k,j] \cdot g[2i-k], \quad HH[i,j] = \sum_{k} I_{\text{high}}[k,j] \cdot h[2i-k].$$

The resulting sub-bands LL, LH, HL, and HH each have dimensions $\frac{n}{2} \times \frac{n}{2}$, capturing the low-low, low-high, high-low, and high-high frequency components, respectively, of the original image.

- *LL*: This sub-band captures the approximation or low-frequency content of the image in both the horizontal and vertical directions. It retains most of the important structural information of the image.
- *LH*: This sub-band captures the vertical high-frequency components (edges and fine details) of the image. It primarily highlights vertical details like vertical edges.
- *HL*: This sub-band captures the horizontal high-frequency components of the image. It emphasizes horizontal edges or features.
- *HH*: This sub-band captures the high-frequency content in both horizontal and vertical directions, corresponding to diagonal details such as corners, intersections, and textures.

These sub-bands collectively represent the image in both spatial and frequency domains. higher values in the LH, HL, and HH sub-bands correspond to regions with higher frequency variations (such as edges or fine details) in the respective subdomains. Higher values in the LL sub-band correspond to regions with higher intensity or brightness in the original image but represent lowfrequency content.

844 C.2 IMPLEMENTATION DETAILS

We implement all models in PyTorch and show their implementation details below.

Bicubic Bicubic interpolation is a widely used image scaling technique that enhances the resolution
of low-resolution images by using a weighted average of pixels in a 4x4 neighborhood around each
pixel. Bicubic interpolation uses 16 neighboring pixels to produce smoother and more accurate
results. This method is deterministic and does not involve any machine learning or training.

851 CNN We modify the hybrid Downsampled Skip-Connection Multi-Scale (DSC/MS) architecture 852 proposed by Fukami et al. (2019). In the original implementation of the DSC/MS model, the model 853 operates on a single frame of velocity field. We modify the model to work in our settings, where 854 the input consists of three consecutive frames of low-fidelity vorticity data, upsampled to $256 \times$ 855 256. The DSC component starts by compressing the input through a series of convolutional layers, effectively capturing multi-scale features. Skip connections are then introduced at various stages to 856 preserve spatial information, ensuring that the network retains fine details from earlier layers. The 857 MS component consists of three parallel convolutional paths that capture features at different scales 858 and combine them to produce the high-resolution output of three consecutive frames of 256×256 859 vorticity data. The training process uses the L2 loss function, running for 300 epochs with a learning 860 rate of 1e - 5. 861

GAN We modify the physics-informed GAN proposed by Li & McComb (2022). In the original implementation of the physics-informed GAN, the input was low-resolution phase fraction data from a multiphase fluid simulation, and the output was a high-resolution phase fraction representation. Our

adaptation uses low-fidelity vorticity data upsampled to 256×256 as input and high-fidelity vorticity data as output. The generator consisted of multiple convolutional layers and residual blocks, while the discriminator had a series of convolutional layers followed by dense layers. The physicsinformed loss function combined a mean squared error (MSE) term with an additional term designed to enforce conservation of mass, ensuring the generated data remained consistent with fluid dynamics principles. The model was trained for 300 epochs with a learning rate of 1e - 5.

870 Diffusion The unconditional diffusion model in Shu et al. (2023) is trained only on high-fidelity 871 CFD data without requiring low-fidelity data during training. The model uses a U-Net architecture 872 with hierarchical convolutional blocks, multi-level skip connections, and a self-attention mechanism 873 in the bottleneck layer to capture complex features of the high-fidelity data. The training process 874 spans 300 epochs, where the model learns to predict and remove noise added during the forward diffusion process. During inference, the model starts with a guided reverse diffusion process from 875 an intermediate timestep t < T instead of starting with complete random Gaussian noise. The low-876 fidelity input data is first upsampled to the target resolution and then combined with Gaussian noise 877 to form the initial input for the reverse diffusion process. This noisy data serves as guidance for 878 generating the high-fidelity output. 879

Conditional Diffusion The conditional diffusion model proposed in Shu et al. (2023) incorporates
 physics-informed guidance during both training and inference by utilizing the gradient of the PDE
 residual with respect to the noised sample. During training, at each step of the forward diffusion
 process, the residual of the governing PDE is computed based on the current noised sample. The
 gradient of this residual is calculated and concatenated with the noised sample as input to the U-Net.
 During inference, the same process is applied: the model computes the gradient of the PDE residual
 at each reverse diffusion step and concatenates it with the noised sample.

887 **PG-Diff** Starting with x_t , which represents an intermediate denoised state at time-step t, the denoising process utilizes a U-Net model to estimate ϵ_{θ} . The UNet architecture starts with an initial 3x3 convolution that maps the in_channels: 3 to ch: 128. The encoder progressively downsamples 889 at each level, using a channel multiplier ch_mult: [1, 1, 1, 2], which means the channels 890 stay at 128 for three levels and then double to 256. Attention blocks are applied at resolutions to 891 capture long-range dependencies. The decoder mirrors the encoder, using upsampling layers to re-892 store spatial resolution and skip connections to preserve details. The output is transformed back to 893 the original out_ch: 3 channels through a final 3x3 convolution. The estimated ϵ_{θ} is then inte-894 grated into a sampling process, outlined in detail by Steps 5 of Algorithm 1, to iteratively generate 895 x_{t-1} . In this approach, the inference phase is governed by several critical hyperparameters: t, which 896 defines where the partial backward diffusion sampling begins; K, the total number of DDIM backward diffusion steps performed; and the set $\{\beta_t\}_{t=1}^T$, which dictates the scaling factors controlling 897 898 noise variance throughout the forward diffusion stages. We adopt the same hyperparameters used in Shu et al. (2023). The training algorithm for our PG-Diff is presented in Algorithm 2. The PG-899 Diff training procedure involves iteratively updating the denoiser ϵ_{θ} using high-fidelity training data 900 $\mathcal{Y}^{\text{train}}$. In each training iteration, a sample y is drawn from the high-fidelity training dataset, and 901 a time-step t is randomly selected from a uniform distribution over $\{1, \ldots, T\}$. Gaussian noise ϵ_t 902 is then sampled from $\mathcal{N}(0, I)$. The high-fidelity data y is transformed into the wavelet domain to 903 compute an importance weight a. The model's parameters θ are updated through a gradient descent 904 step using the gradient $\nabla_{\theta} a \odot \| \epsilon_t - \epsilon_{\theta} (\sqrt{\overline{\alpha}_t} y + \sqrt{1 - \overline{\alpha}_t} \epsilon_t, t) \|^2$. This process repeats until the model 905 converges, enabling the denoiser to learn to reconstruct fine-grained details in the high-fidelity data 906 effectively.

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D DATASET

All datasets are generated using pseudo-spectral solver implemented by Li et al. (2020). Our dataset
will be released upon publication of our paper and be free to use.

The time-stepping method employed is a combination of the Crank-Nicholson scheme and Heun's method. The Crank-Nicholson scheme is an implicit, second-order accurate method, and it is applied to the viscous term. This allows the solver to be stable even for relatively large time steps when handling viscous diffusion. Heun's method, a second-order Runge-Kutta technique, is used to handle the non-linear advection term. The combination of these two methods provides an efficient and

918 Algorithm 2 PG-Diff Model Training. 919 **Require:** $\mathcal{Y}^{\text{train}} \sim p_{\mathcal{Y}}^{\text{train}}$ (High-fidelity data used for training), ϵ_{θ} (Denoiser to be trained) 920 1: repeat 921 $y \in \mathcal{Y}^{\mathrm{train}}$ 2: 922 3: $t \sim \text{Uniform}(\{1, \ldots, T\})$ 923 4: $\epsilon_t \sim \mathcal{N}(0, I)$ 924 5: Transform y into the wavelet domain and compute the importance weight a925 6: Take a step of gradient descent on θ with the following gradient: 926 $\nabla_{\theta} a \odot \| \epsilon_t - \epsilon_{\theta} (\sqrt{\bar{\alpha}_t} y + \sqrt{1 - \bar{\alpha}_t} \epsilon_t, t) \|^2$ 927 928 7: **until** converged 929 930 931 accurate way to evolve the vorticity field over time, with the implicit Crank-Nicholson step ensuring 932 stability for stiff viscous terms, and Heun's method capturing the non-linearity of the advection. 933

For numerical stability, the solver uses an adaptive time-stepping approach governed by the Courant-Friedrichs-Lewy (CFL) condition. The CFL condition ensures that the time step remains sufficiently small relative to the velocity field and the external forcing, preventing instabilities that can arise from rapid changes in the solution. Additionally, the dealiasing procedure, using the 2/3 rule, removes high-frequency components from the Fourier spectrum, ensuring that non-physical aliasing effects are avoided.

Algorithm 3 Pseudo-spectral Navier-Stokes Solver

941 1: Input: Initial vorticity ω_0 , forcing f, Reynolds number Re, total time T, timestep Δt , domain 942 size L_1 , L_2 , grid size s_1 , s_2 , adaptivity flag 943 **Output:** Vorticity ω at time T 3: Initialize: Compute wavenumbers, Laplacian, and dealiasing mask 944 4: 945 5: while t < T do 946 if adaptive then 6: 947 7: Compute velocity field $\boldsymbol{u} = \nabla^{\perp} \boldsymbol{\psi}$ 948 8: Update timestep Δt based on CFL 949 9: end if 950 10: Compute non-linear term in Fourier space 951 11: Predictor and corrector steps using Crank-Nicholson + Heun 952 12: Apply dealiasing mask 953 13: Update time $t = t + \Delta t$ 14: end while 954 15: **return** Vorticity ω 955

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957 **Taylor Green Vortex** The initial vorticity field is based on the analytical solution of the TGV, and to 958 generate different trajectories, we added random perturbations from a Gaussian random field. These 959 perturbations introduce variability to the initial conditions while maintaining the overall vortex struc-960 ture. No external forcing was applied during the simulation, and the spatial domain is $[0, \frac{3}{2}\pi]^2$ with a 961 fixed Reynolds number of 1000. The simulation used a time step $dt = \frac{1}{32}$, and 100 trajectories were 962 generated each with a total duration of T = 6 seconds. The initial vorticity field is based on the 963 analytical solution for the two-dimensional periodic domain. The vorticity field ω is initialized as $\omega = -2U_0 k \sin(kx) \sin(ky)$, where U_0 is the initial velocity amplitude, and k is the wave number 964 that determines the size of the vortices. To introduce variability and generate different trajectories, 965 a Gaussian Random Field is added as a perturbation to the initial vorticity. 966

967 **Decaying Turbulence** The spatial domain is $[0, 1]^2$, with periodic boundary conditions and a fixed 968 Reynolds number of 450. The simulation used a time step $dt = \frac{1}{32}$, and 400 trajectories were 969 generated, each with a total duration of T = 2 seconds. The initial conditions for the decaying 970 turbulence dataset are generated by superimposing randomly positioned vortices of varying intensity 971 and size. Each vortex is characterized by a randomly selected core size and maximum rotational 962 velocity, allowing for a diverse range of initial flow structures. The vortices are distributed randomly throughout the domain, and their periodic images are added to ensure the proper enforcement of periodic boundary conditions.

Kolmogorov Flow The initial vorticity field is generated using a Gaussian random field, and the 975 system is subjected to a forcing term of the form $f(x) = -4\cos(4x_2) - 0.1\omega(x, t)$. This forcing 976 drives the flow in the y-direction while introducing a drag force that dissipates energy. The spatial 977 domain is $[0, 2\pi]^2$, with a fixed Reynolds number of 1000. The simulation used a time step $dt = \frac{1}{32}$, 978 and 50 trajectories were generated, each with a total duration of T = 10 seconds. The initial vorticity 979 field is produced by sampling from a Gaussian Random Field. As the external forcing continually 980 adds energy to the system, the initially simple vorticity evolves into intricate and turbulent structures. 981 The vorticity is allowed to evolve over a 5-second period, and the final state at the end of this interval 982 is used as the initial condition for our dataset.

983 McWilliams Flow The phenomenon illustrates the emergence of order from initially disordered 984 turbulent motion, driven by viscous dissipation and the self-organization of the flow. No external 985 forcing is applied during the simulation, allowing for a natural decay and evolution of the turbulence. 986 The spatial domain is $[0, 2\pi]^2$, with periodic boundary conditions and a Reynolds number of 2000, 987 providing a high degree of turbulence. The simulation used a time step $dt = \frac{1}{32}$, and 50 trajectories were generated, each with a total duration of T = 10 seconds. The initial vorticity field for the 988 989 *McWilliams Flow* is generated following the method described by Mcwilliams (1984). The process begins by constructing a Fourier mesh over the spatial domain, where the wavenumbers k_x and k_y 990 are calculated. A scalar wavenumber function is prescribed, and the ensemble variance is determined 991 to ensure that the energy distribution in Fourier space follows the desired spectral shape. Random 992 Gaussian perturbations are applied to each Fourier component of the stream function, producing a 993 random realization of the vorticity field. To ensure the stream function has a zero mean, a spectral 994 filter is applied, and the field is normalized based on the kinetic energy. Finally, the vorticity field is 995 computed in physical space by taking the inverse Laplacian of the stream function in Fourier space, 996 resulting in a turbulent flow field that evolves naturally without external forcing. 997

We show the visualization of the four datasets below.



Figure 5: Example trajectory of the Taylor Green Vortex dataset. The first row is the high-fidelity data with 256×256 discretization grid. The second and third rows are low fidelity data with 64×64 and 32×32 discretization grids, respectively. For visualization, we upsample the low-fidelity data to 256×256 discretization grid.

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1020 1021 D.1 RESIDUAL CALCULATION

The PDE residual of the governing equation $\mathcal{R}(\omega)$ is defined as

1024 1025 $\mathcal{R}(\omega) = rac{1}{N} \sum_{i,j} \left[\text{LHS}(\omega)_{i,j} - \text{RHS}(\omega)_{i,j}
ight]^2$



Figure 6: Example trajectory of the Decaying Turbulence dataset. The first row is the high-fidelity data with 256×256 discretization grid. The second and third rows are low fidelity data with 64×64 and 32×32 discretization grids, respectively. For visualization, we upsample the low-fidelity data to 256×256 discretization grid.



Figure 7: Example trajectory of the Kolmogorov Flow dataset. The first row is the high-fidelity data with 256×256 discretization grid. The second and third rows are low fidelity data with 64×64 and 32×32 discretization grids, respectively. For visualization, we upsample the low-fidelity data to 256×256 discretization grid.



Figure 8: Example trajectory of the *McWilliams Flow* dataset. The first row is the high-fidelity data with 256×256 discretization grid. The second and third rows are low fidelity data with 64×64 and 32×32 discretization grids, respectively. For visualization, we upsample the low-fidelity data to 256×256 discretization grid.

where $LHS(\omega)$ and $RHS(\omega)$ represent the left-hand side and right-hand side expressions of the PDE, respectively. ω represents the vorticity. N represents the number of grid points. For the incompressible navier stokes equation tested in this paper, the residual is defined as

$$\mathcal{R}(\omega) = \frac{1}{N} \sum_{i,j} \left[\frac{\partial \omega(\boldsymbol{x},t)}{\partial t} + \boldsymbol{u}(\boldsymbol{x},t) \cdot \nabla \omega(\boldsymbol{x},t) - \frac{1}{Re} \nabla^2 \omega(\boldsymbol{x},t) - f(\boldsymbol{x}) \right]^2$$

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1108 E ADDITIONAL EXPERIMENTAL RESULTS

1110 E.1 DWT VISUALIZATION

Figure 9 presents sample importance masks identified as high-frequency features using the DWT. These masks highlight the regions where our model focuses to capture intricate details. The samples clearly demonstrate that our method effectively captures the fine-grained structures in the turbulent flow, validating its ability to reconstruct small-scale vortices and turbulence that are often missed by conventional approaches.



Figure 9: High frequenct components in the *Kolmogorov Flow* field identified by discrete wavelet transformation. We compare the high fidelity vorticity field with the importance mask.

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1130 E.2 MULTI-SCALE EVALUATION

To comprehensively evaluate the model's ability to capture both overall flow dynamics and intricate details, we performed a multi-scale analysis by transforming the predicted and ground truth flow fields into the wavelet domain. This transformation produced four subdomains: LL (low-low), LH (low-high), HL (high-low), and HH (high-high). The LL subdomain represents broad, low-frequency components, while the LH, HL, and HH subdomains capture increasingly finer, high-frequency features, including turbulent structures. We then assessed the L2 norm across these sub-domains to gain deeper insights into the model's effectiveness in reconstructing both large-scale patterns and fine-grained details.



Figure 10: Wavelet subdomain L2 for the Taylor Green Vortex dataset.



Figure 11: Wavelet subdomain L2 for the Decaying Turbulence dataset.

1184 E.3 RECONSTRUCTION VISUALIZATION

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We present visualizations of the reconstructed high-fidelity data using PG-DIFF w/o Cor across
 different methods and datasets. As shown in Figure 14 and Figure 15, PG-Diff consistently produces impressive qualitative results.





Figure 14: Visualization comparison of reconstruction on the Kolmogorov Flow from PG-Diff w/o Cor. The first column displays the low-fidelity input data upscaled to a resolution of 256×256 , 1268 while the last column shows the high-fidelity ground truth. 1269

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1271 E.4 LPIPS SCORES 1272

1273 Direct mapping models are effective at minimizing L2 loss. However, these models tend to produce 1274 samples that appear smooth and blurry, often missing the intricate details. This is because the 1275 L2 loss emphasizes pixel-wise accuracy, which can lead to averaging effects and the loss of fine-1276 grained details. To address this issue, we use the LPIPS (Learned Perceptual Image Patch Similarity) 1277 metric, which focuses on perceptual differences and provides a more qualitative assessment of the 1278 reconstructed flow fields, capturing the texture and small-scale structures. The results are presented in Table 4 1279

1280 Although LPIPS is trained on the ImageNet dataset, which consists of natural images, it remains a 1281 valuable metric for evaluating perceptual quality in CFD applications. This is because LPIPS lever-1282 ages features from deep neural networks that are effective at capturing multi-scale patterns, textures, 1283 and perceptual similarities, regardless of the specific domain. Fluid dynamics data often have com-1284 plex structures and turbulent patterns that share characteristics with textures found in natural images, making LPIPS suitable for assessing the fidelity of reconstructed flow fields. Thus, despite being 1285 trained on ImageNet, LPIPS can still effectively quantify how well the reconstructed samples re-1286 tain important perceptual details, making it a robust metric for evaluating the visual quality of CFD 1287 reconstructions. 1288

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1290 E.5 IMPORTANCE WEIGHT SENSITIVITY ANALYSIS

We present a sensitivity analysis to explore how different hyperparameter settings affect the performance of the Importance Weight Strategy. The importance weight strategy is governed by three 1293 main parameters: β (the maximum importance weight), α (the minimum importance weight), and θ 1294 (the importance threshold). We adjust these parameters sequentially. The results are summarized in 1295 Figure 16.



Figure 15: Visualization comparison of reconstruction on the *McWilliams Flow* from PG-Diff w/o Cor. The first column displays the low-fidelity input data upscaled to a resolution of 256×256 , while the last column shows the high-fidelity ground truth.

	Bicubic	CNN	GAN	Diff	Cond Diff	PG-Diff
Kolmogorov	0.5421	0.5358	0.5109	0.2848	0.2869	0.2781
Flow	0.2919	0.4065	0.4082	0.1215	0.1229	0.1097
McWilliams	0.5736	0.4921	$0.4505 \\ 0.3948$	0.3524	0.3540	0.2936
Flow	0.3417	0.3948		0.1457	0.1604	0.1298
Decaying	0.2221	0.1448	0.1991	0.4215	0.1688	0.1397
Turbulence	0.0794	0.1521	0.1407	0.3608	0.2259	0.0637
Taylor Green	0.3715	0.3331	0.4175	0.2525	0.1750	0.1704
Vortex	0.1594	0.3057	0.3137	0.1494	0.2362	0.1339

Table 4: LPIPS scores for each dataset. Metrics are reported for both $32 \times 32 \rightarrow 256 \times 256$ (grey) and $64 \times 64 \rightarrow 256 \times 256$ tasks.

E.6 COMPARISON OF RESIDUAL CORRECTION AND POST PROCESSING

We compared the performance of residual correction and post-processing technique in Table 5. The
post-processing applies a single residual correction to the reconstructed high-fidelity samples in the
end. The results indicate that while post-processing achieves very low PDE residuals, it significantly
increases the L2 loss. In contrast, our Residual Correction method strikes a better balance between
L2 loss and PDE residual.



Figure 16: Sensitivity analysis of key parameters for the *Importance Weight*. Experiments are conducted on the *Kolmogorov Flow*. The top row presents the results for the maximum importance weight, β . The middle row displays the results for the minimum importance weight, α , and the bottom row shows the results for the importance threshold, θ . These three hyperparameters were tuned in sequence, and the optimal combination ($\beta = 6$, $\alpha = 1.25$, and $\theta = 0.8$) is selected.

1404		$32 \times 32 \rightarrow$	256×256	$64 \times 64 \rightarrow$	256×256
1405		L2	PDE Residual	L2	PDE Residual
1406	Residual Correction	2.7897	40.12	1.6609	39.97
1407	Post Processing	3.0019	5.42	1.7737	7.16
1408					

Table 5: Comparison between our proposed Residual Correction and Post Processing methods. Experiments are conducted on the *Kolmogorov Flow*.

1413 E.7 GENERALIZATION

We show the generalization ability of our model across various settings. The original *Kolmogorov Flow* dataset is generated using timestep dt = 1/32, spatial domain size $2\pi \times 2\pi$, and Reynolds number Re = 1000.

Table 6: Generalization results on *Kolmogorov Flow* dataset with $64 \times 64 \rightarrow 256 \times 256$ setting.

Variation	Model	L2	PDE Residual			
Time Discretization Variations						
dt = 1/40	Trained on Original Data	1.6782	31.49			
	Trained on $dt = 1/40$ Data	1.6/23	30.91			
dt = 1/50	Trained on Original Data	1.6484	33.49			
	Trained on $dt = 1/50$ Data	1.6489	26.66			
Spatial Domain Size Variations						
Domain Size $1\pi \times 1\pi$	Trained on Original Data	1.0255	263.23			
	Trained on $1\pi \times 1\pi$ Data	0.9443	207.03			
Domain Size $1.5\pi \times 1.5\pi$	Trained on Original Data	1.3212	84.24			
	Trained on $1.5\pi \times 1.5\pi$ Data	1.2918	94.87			
	Reynolds Number Variations					
Re = 500	Trained on Original Data	1.3112	21.57			
	Trained on $Re = 500$ Data	1.2998	22.96			
Re = 2000	Trained on Original Data	1.9647	24.07			
	Trained on $Re = 2000$ Data	1.9694	24.22			

E.8 RUNTIME COMPARISON

1443Table 9 presents the time required for the numerical solver to generate a single frame of low- and1444high-fidelity samples for each dataset. The results indicate that generating even 64×64 low-fidelity1445data is significantly faster than producing high-fidelity data, emphasizing the value of using ML1446approach for accelerating high-fidelity simulations.

Dataset	256×256	64×64	32×32
Kolmogorov Flow	138.49	0.44	0.04
McWilliams Flow	98.99	1.23	0.12
Decaying Turbulence	93.82	1.01	0.06
Taylor Green Vortex	139.35	1.49	0.22

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1454Table 7: Run time of numerical solver to generate one frame for each dataset across different grid
resolutions with a batch size of 10. All times are measured in seconds.

1457 Table 8 shows the time comparison across ML models for reconstructing high-fidelity data from low-fidelity inputs.

1458	Method	$32 \times 32 \rightarrow 256 \times 256$	$64 \times 64 \rightarrow 256 \times 256$
1459	Method	32,32 7 230,230	01/01 / 250/250
1460	Bicubic	1.43e-5	1.4e-5
1461	CNN	0.005	0.005
1460	GAN	0.008	0.008
1402	Diff	6.10	3.06
1463	Cond Diff	6.20	3 38
1464	PG-Diff	6.37	3 27
1465		0.57	5.21
1466	DC Diff w/a Car	6 10	2.06
1467	PC Diff w/s UV	0.10	3.00
1468	PG-Dill W/0 IW	0.37	5.27

1469Table 8: Runtime comparison of various methods across different resolution levels using a batch1470size of 10, with all times measured in seconds. These results are based on the *McWilliams Flow*1471dataset.

Dataset	Same error with 64×64	Same error with 32×32
Kolmogorov Flow	35.36	15.15
McWilliams Flow	39.08	17.81
Decaying Turbulence	21.51	17.41
Taylor Green Vortex	33.23	15.49

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1479Table 9: Run time of numerical solver to generate one frame for each dataset with approximately
same error compared to PG-Diff with a batch size of 10. All times are measured in seconds.

1482 E.9 EMBEDDING SUPER RESOLUTION WITHIN SOLVER

For longer rollouts on coarser grids, low-fidelity data becomes qualitatively different from highfidelity data, making simple postprocessing with super-resolution models insufficient to recover the trajectory. However, we demonstrate that when integrated into the solver, PG-Diffenhances numer-ical simulations on coarse grids. Our approach follows the pipeline: "solver \rightarrow super resolution \rightarrow downsample \rightarrow solver." Specifically, we begin with a numerical solver to perform one-step predic-tions on a coarse grid. Next, we apply PG-Diff for super-resolution, downsample the output back to the coarse grid, and use it as input for the next simulation step. We present the visual results in Figure 17.

E.10 PSNR AND SSIM RESULTS

Table 10: Kolmogorov Flow

Model	4x Upsa	mpling	8x Upsampling	
	PSNR	SSIM	PSNR	SSIM
Bicubic	21.1257	0.4769	18.3063	0.2479
CNN	24.8310	0.5190	22.9806	0.3712
GAN	20.6210	0.4160	20.2323	0.3695
Diffusion	25.4049	0.6487	21.5818	0.4072
Conditional Diffusion	25.2389	0.6456	20.2067	0.3425
PG-Diff	26.1733	0.6781	24.0754	0.4409



Figure 17: Visualization of reconstructed high-fidelity data when PG-Diffis integrated within the numerical solver.

Table 11: McWilliams Flow

Model	4x Upsa	mpling	8x Upsampling	
	PSNR	SSIM	PSNR	SSIM
Bicubic	25.1992	0.4834	21.9313	0.2519
CNN	28.6248	0.5897	27.4487	0.5164
GAN	28.6720	0.4621	27.7062	0.4477
Diffusion	29.71018	0.6686	25.2200	0.3884
Conditional Diffusion	29.6757	0.6665	25.1475	0.3823
PG-Diff	30.0540	0.6722	28.1972	0.4643

Table 12: Taylor Green Vortex

Model	4x Upsampling		8x Upsampling	
	PSNR	SSIM	PSNR	SSIM
Bicubic	25.4811	0.7211	19.4033	0.5041
CNN	26.6066	0.7657	26.3448	0.7422
GAN	28.1370	0.7597	25.0551	0.7633
Diffusion	30.7706	0.8438	24.8698	0.6678
Conditional Diffusion	23.2154	0.6767	21.3979	0.5959
PG-Diff	31.5277	0.8713	26.7503	0.7562

Table 13: Decaying Turbulence

Model	4x Upsampling		8x Upsampling	
	PSNR	SSIM	PSNR	SSIM
Bicubic	41.7820	0.8888	35.1032	0.7098
CNN	41.0336	0.8849	42.1580	0.9200
GAN	39.2718	0.9055	38.9931	0.8863
Diffusion	47.2289	0.9571	40.8901	0.8622
Conditional Diffusion	46.0675	0.9138	40.1089	0.8456
PG-Diff	48.0321	0.9601	43.0276	0.9295